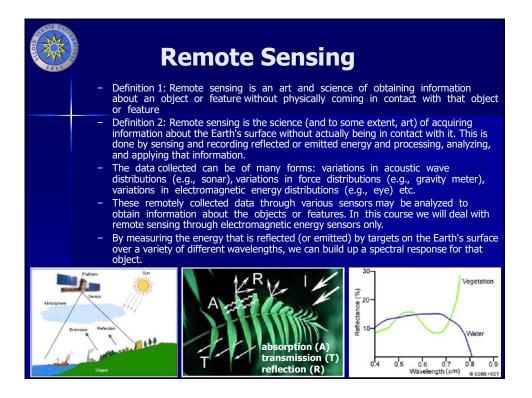
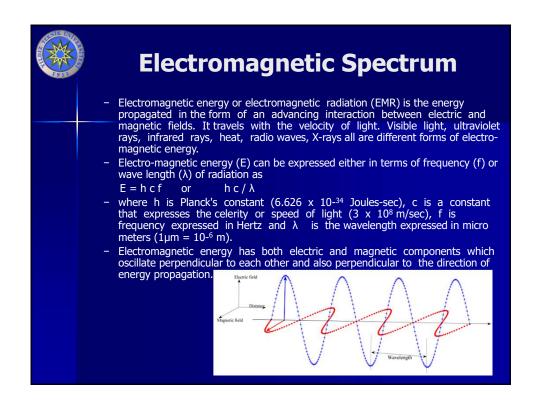
REMOTE SENSING SENSORS

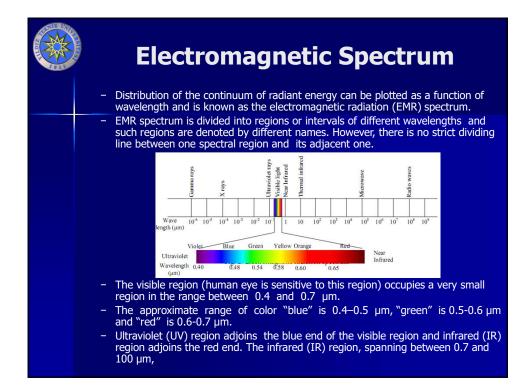
Prof. Dr. Naci YASTIKLI

Yıldız Technical University Department of Geomatic Engineering

İstanbul 2019







Region	Wavelength (µm)	Remarks
Gamma rays	< 3×10 ⁻⁵	Not available for remote sensing. Incoming radiatio is absorbed by the atmosphere
X-ray	3×10*5 - 3×10*3	Not available for remote sensing since it is absorbed by atmosphere
Ultraviolet (UV) rays	0.03 - 0.4	Wavelengths less than 0.3 are absorbed by the ozonu- layer in the upper atmosphere. Wavelengths between 0.3- 0.4 µm are transmitted and termed are "Photographic UV band".
Visible	0.4 - 0.7	Detectable with film and photodetectors.
Infrared (IR)	0.7 - 100	Atmospheric windows exist which allows maximun transmission. Portion between 0.7 and 0.9 µm i called photographic IR band, since it is detectable with film. Two principal atmospheric windows exis in the thermal IR region (3 - 5 µm and 8 - 14 µm).
Microwave	10 ³ - 10 ⁶	Can penetrate rain, fog and clouds. Both active and passive remote sensing is possible. Radar use

Radio

 $> 10^{6}$

n

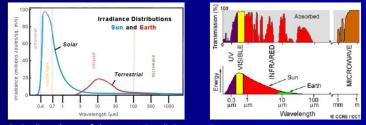
Electromagnetic Spectrum

wavelength in this range.

sensing by some radars.

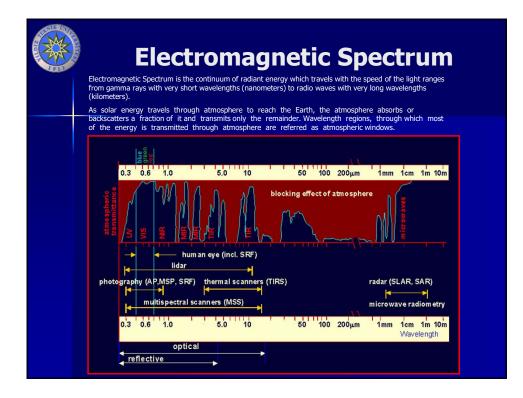
Have the longest wavelength. Used for remote

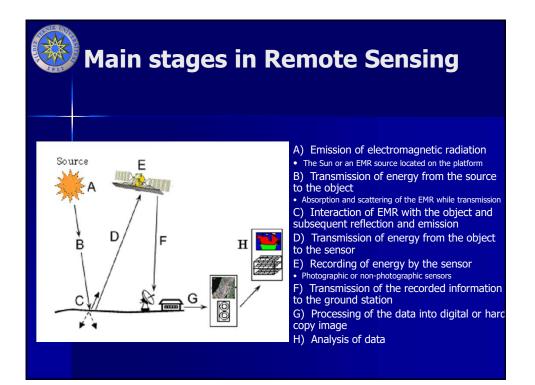
- Primary source of energy that illuminates different features on the earth surface is the Sun but it is not the only one. the Earth and the terrestrial objects also are the sources of electromagnetic radiation. All matter at temperature above absolute zero (0°K or -273°C) emits electromagnetic radiations continuously.
- Although the Sun produces electromagnetic radiation in a wide range of wavelengths, the amount of energy it produces is not uniform across all wavelengths.



The solar irradiance (power of electromagnetic radiation per unit area incident on a surface) distribution of the Sun and Earth.

- Almost 99% of the solar energy is within the wavelength range of 0.28-4.96 μ m. Within this range, 43% is radiated in the visible wavelength region between 0.4-0.7 μ m. The maximum energy (E) is available at 0.48 μ m wave length, which is in the visible green region.





Characteristics of Remote Sensing Systems

Energy Source

The energy sources for real systems are usually non-uniform over various wavelengths and also vary with time and space. This has major effect on the passive remote sensing systems. The spectral distribution of reflected sunlight varies both temporally and spatially. Earth surface materials also emit energy to varying degrees of efficiency. A real remote sensing system needs calibration for source characteristics.

The Atmosphere

- The atmosphere modifies the spectral distribution and strength of the energy received or emitted. The effect of atmospheric interaction varies with the wavelength associated, sensor used and the sensing application. Calibration is required to eliminate or compensate these atmospheric effects. To record reflected energy from object, remote sensing sensors are deployed (mounted) on platforms. Remote sensing platforms can be classified based on the elevation from the Earth's surface at which these platforms are placed.
- The Energy/Matter Interactions at the Earth's Surface
 - Remote sensing is based on the principle that each and every material reflects or emits energy in a unique, known way. However, spectral signatures may be similar for different material types. This makes differentiation difficult. Also, the knowledge of most of the energy/matter interactions for earth surface features is either at elementary level or even completely unknown.

Characteristics of Remote Sensing Systems

The Sensor

Real sensors have fixed limits of spectral sensitivity i.e., they are not sensitive to all wavelengths. Also, they have limited spatial resolution (efficiency in recording spatial details). Selection of a sensor requires a trade-off between spatial resolution and spectral sensitivity. For example, while photographic systems have very good spatial resolution and poor spectral sensitivity, non-photographic systems have poor spatial resolution.

The Data Handling System

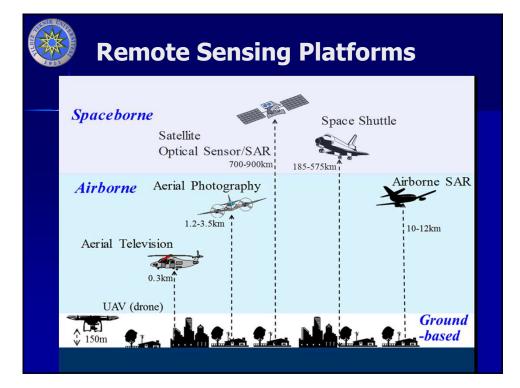
 Human intervention is necessary for processing sensor data; even though machines are also included in data handling. This makes the idea of real time data handling almost impossible. The amount of data generated by the sensors far exceeds the data handling capacity.

The Multiple Data Users

 The success of any remote sensing mission lies on the user who ultimately transforms the data into information. This is possible only if the user understands the problem thoroughly and has a wide knowledge in the data generation. The user should know how to interpret the data generated and should know how best to use them.



- Remote Sensing sensors are deployed (mounted) on platforms to record reflected energy from object.
- Remote sensing platforms can be classified as follows based on the platform elevation from the Earth's surface;
 - Ground (land) platforms: up to 50 m
 - Airborne based platforms: up to 50 km
 - Space shuttles: 250-300 up to km
 - Space stations : 300-400 up to km
 - Satellites
 - Low level satellites: up to 500-2000 km
 - Middle level satellites: up to 10000 km
 - High level satellites: up to 36000 km
 - The remote sensing platform affects the image resolution and image coverage on the ground.





Ground based platforms

They are basically used for verification or calibration of the data collected from sensor mounted on airborne based platforms, space shuttles, space stations, and satellites. Ground based platform can move or fixed. The Spectral reflectance meters can be used by hand to identify the reflectance characteristics of individual leafs, plants or areas.





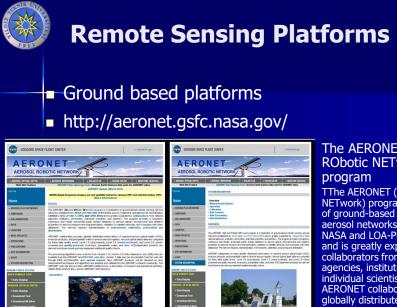
Remote Sensing Platforms

Ground based platforms http://www.arm.gov/



Atmospheric Radiation Measurement (ARM)

The ARM Climate Research Facility, a DOE scientific user facility, provides the climate research community with strategically located in situ and remote sensing observatories designed to improve the understanding and representation, in climate and earth system models, of clouds and aerosols as well as their interactions and coupling with the Earth's surface.



The AERONET (AErosol RObotic NETwork)

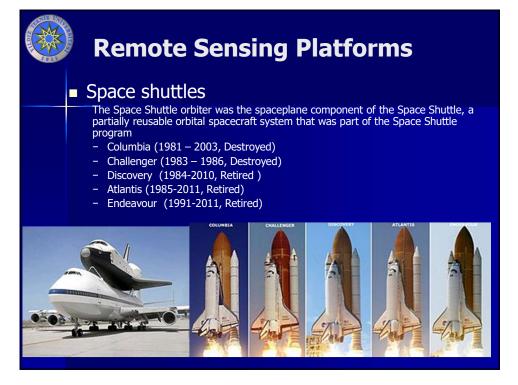
program TThe AERONET (AErosol RObotic NETwork) program is a federation of ground-based remote sensing aerosol networks established by NASA and LOA-PHOTONS (CNRS) and is greatly expanded by collaborators from national agencies, institutes, universities, individual scientists, and partners. AERONET collaboration provides globally distributed observations of spectral aerosol optical Depth (AOD), inversion products, and precipitable water in diverse aerosol regimes. aerosol regimes.



Space shuttles

- The Space Shuttle was a partially reusable human spaceflight vehicle capable of reaching low Earth orbit, commissioned and operated by the U.S. National Aeronautics and Space Administration (NASA) from 1981 to 2011.
- The official program name was Space Transportation System (STS)
- Operational missions launched numerous satellites, interplanetary probes, and the Hubble Space Telescope (HST); conducted science experiments in orbit; and participated in construction and servicing of the International Space Station.
- The Shuttle fleet's total mission time was 1322 days, 19 hours, 21 minutes and 23 seconds
- Shuttle components included the Orbiter Vehicle (OV) with three clustered Rocketdyne RS-25 main engines, a pair of recoverable solid rocket boosters (SRBs), and the expendable external tank (ET) containing liquid hydrogen and liquid oxygen.
- The Space Shuttle was launched vertically, like a conventional rocket, with the two SRBs operating in parallel with the OV's three main engines, which were fueled from the ET.





Space stations

- The space stations are research platforms, used to study the effects of long-term space flight on the human body as well as to provide platforms for greater number and length of scientific studies than available on other space vehicles.
- The first space station was Salyut 1, which was launched by the Soviet Union on April 19, 1971.
- The earlier Soviet stations were all designated "Salyut", but among these there were two distinct types: civilian and military. The military stations, Salyut 2, Salyut 3, and Salyut 5, were also known as Almaz stations.
- The civilian stations Salyut 6 and Salyut 7 (1986) were built with two docking ports, which allowed a second crew to visit, bringing a new spacecraft with them







Remote Sensing Platforms

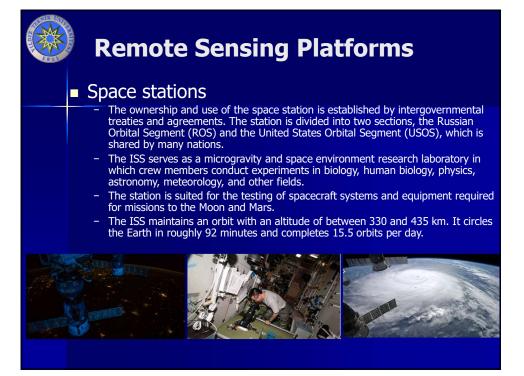
Space stations

- The Soviet space station Mir (1986-2001) had a modular design; a core unit was launched, and additional modules, generally with a specific role, were later added to that. This method allows for greater flexibility in operation, as well as removing the need for a single immensely powerful launch vehicle.
- Skylab was a United States space station launched and operated by NASA, and occupied for about 24 weeks between May 1973 and February 1974 – the only space station the U.S. has operated exclusively

Mission	Emblem	Commander	Science Pilot	Pilot	Launch date	Landing date	Duration (days)
Skylab 1 SL-1	-	unmanne	id launch of spa	ce station	1973-05-14 17:30:00 UTC	1979-07-11 16:37:00 UTC	2248.96
Skylab 2 SL-2 (SLM- 1)	3	Pete Corrad	Joseph Kerwin	Paul Weitz	1973-05-25 13:00:00 UTC	1973-06-22 13:49:48 UTC	28.03
Skylab 3 SL-3 (SLM- 2)		Alan Bean	Owen Garriott	Jack Lousma	1973-07-28 11:10:50 UTC	1973-09-25 22:19:51 UTC	59.46
Skylab 4 SL-4 (SLM- 3)	Ø	Gerald Carr	Edward Gibson	William Pogue	1973-11-16 14:01:23 UTC	1974-02-08 15:16:53 UTC	84.04
Skylab 5		Vance Brand	William B. Lenoir	Don Lind	(April 1974, Cancelled)		20 (notional)
Skylab Rescue		Vance Brand	Don Lind	(Survivors)	(On Standby)		

Account of the second se







Satellites

Satellite classification by mass criterions

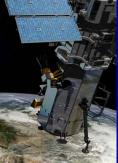
- Large satellite: > 1000 kg (Hubble, Türksat 3A) Medium satellites: 500 kg 1000 kg (THEOS) Mini satellites: 100 kg 500kg (Göktürk -II) WorldView-3 Micro satellites: 10 kg 100 kg (UoSAT-1) On August 13 Nano satellites: 1 kg 10 kg (Delfi C3) Piko satellites: 0,1kg 1kg (<TÜpSAT1) Femto satellites: ağırlığı 0,1 kg dan az 1.24 m multis
- On August 13, 2014, DigitalGlobe launched WorldView-3 into orbit 31 cm pankromatik

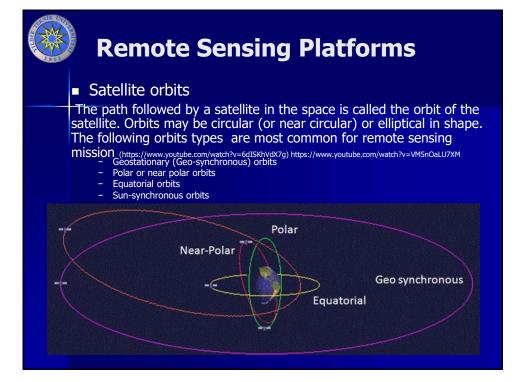
- Satellite classification by application areas
- Military satellite
- Communication satellite
- meteorological satellite
- Navigation satellite Earth observation satellite
- Space exploration satellite

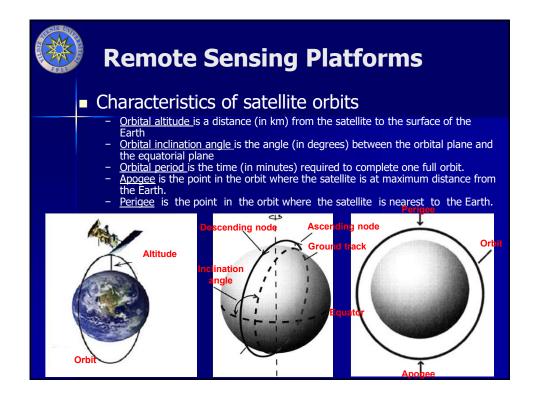
Satellite classification by orbit types

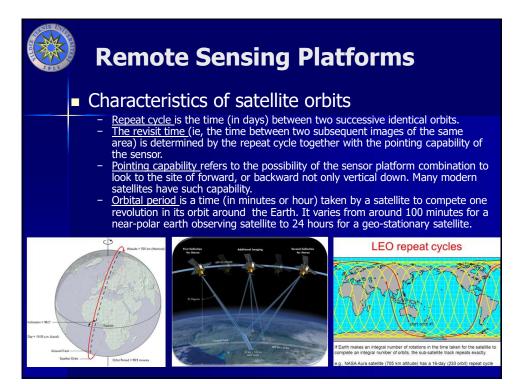
- (On the basis of the distance from Earth)
 Low Earth Orbit (LEO) satellite: 500-2000 km
 Medium Earth Orbit (MEO) satellite: 10000 km
 High Earth Orbit: (HEO) satellite: 36000 km

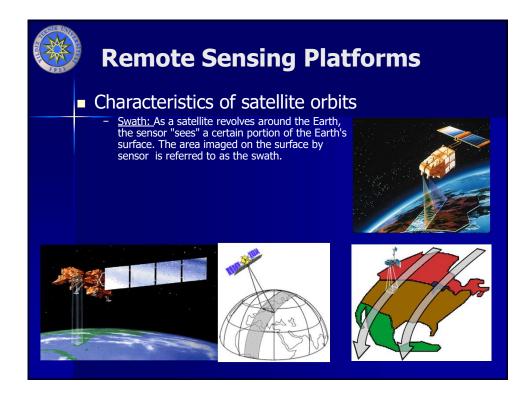


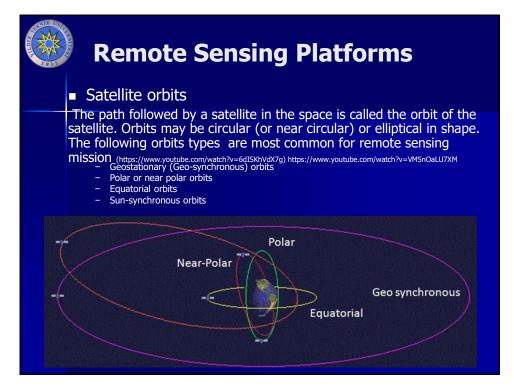






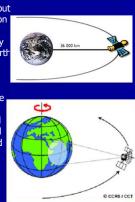






Geostationary (Geo-synchronous) orbits

- Although terms are often used interchangeably, technically a geosynchronous orbit matches the Earth's rotational period, but the definition does not require it to have zero orbital inclination to the equator.
- Geosynchronous orbit (GSO)'s inclination and eccentricity may not necessarily be zero. In this case satellite may oscillate north and south during the course of a day. Thus, a geostationary orbit is defined as a geosynchronous
- orbit at zero inclination.
- Geostationary orbit (GEO) is an orbit which places the satellite
- above the same location at all times. Geosynchronous orbit (GSO) period around the Earth is equal to one sidereal day, which is Earth's average rotational period of 23 hours, 56 minutes, 4.091 seconds. They must be placed at a very high altitude approximately 36,000 km in order to produce an orbital period equal to the period of Earth's rotation.
- All geostationary orbits must be Geo-synchronous, but not all Geo-synchronous orbits are Geostationary.
- Geostationary or geosynchronous orbits are located in the equatorial plane and used for communication and meteorological satellites. Example: INSAT, MeteoSAT, GOES, GMS etc.

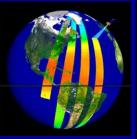


Remote Sensing Platforms

Polar or near polar orbits

- Polar orbit passes above or nearly above both North and South poles of the Earth on each revolution.
- Therefore, it has an inclination of (or very close to) 90 degrees.
- The near polar orbit is an orbit with an inclination angle between 80 degrees and 100 degrees.
- Polar orbits or near polar orbits are usually medium or low orbits (approximately 700-800km) compared to the geo-synchronous orbits. Consequently the orbit period is less, which typically varies from 90-103 minutes.
- The satellites in the polar orbits make more than one revolution around the earth in a single day and enables observation of the whole globe also near poles.
- A polar orbiting satellite eventually sees every part of the Earth's surface, which is highly desirable for remote sensing applications.
- Most of the remote sensing satellite platforms today are in near-polar orbits, which means that the satellite travels northwards on one side of the Earth and then toward the southern pole on the second half of its orbit. These are called ascending and descending passes, respectively.





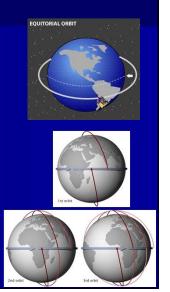
Equatorial orbits

- A satellite in equatorial orbit flies along the line of the Earth's equator.
- To get into equatorial orbit, a satellite must be launched from a place on Earth close to the equator.
 - NASA often launches satellites aboard an Ariane rocket into equatorial orbit from French Guyana.
- Equatorial orbits can be useful for satellites observing tropical weather patterns, as they can monitor cloud conditions around the globe.
- Equatorial orbits are usually medium or low orbits.

Sun-synchronous orbits

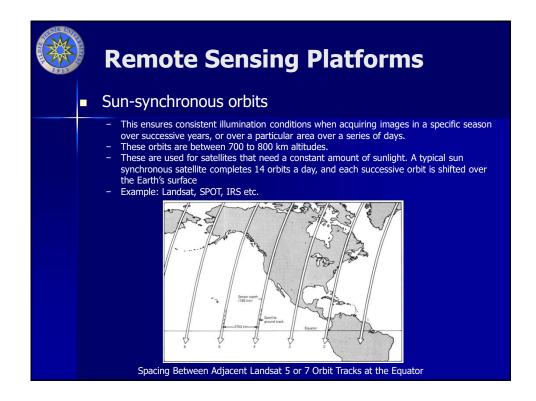
https://www.youtube.com/watch?v=tOp1UYbmp0Y https://www.youtube.com/watch?v=nrtsd8Wxmw0

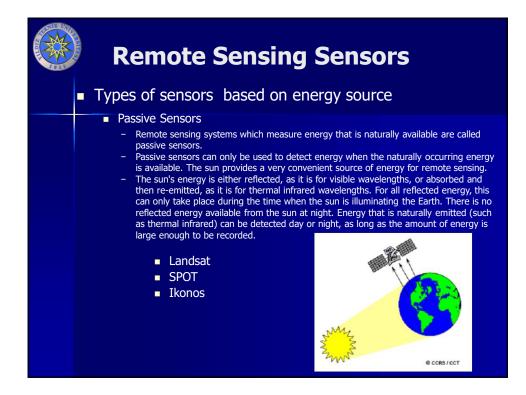
- The sun-synchronous orbit is a special case of polar or near polar orbit.
 - Like a polar orbit, the satellite travels from the north to the south poles as the Earth turns below it. In a sup-synchronous orbit, the satellite passes
- In a sun-synchronous orbit, the satellite passes over the same part of the Earth at roughly the same local time (mid-morning around 10.30) each day.

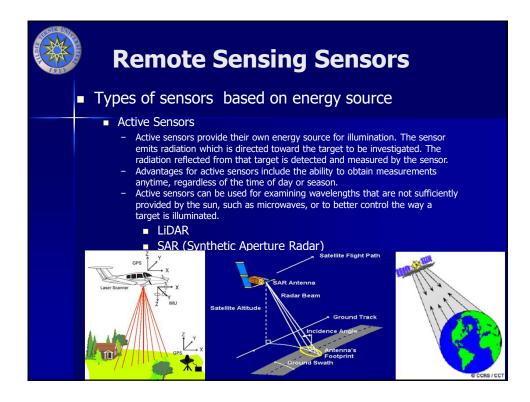


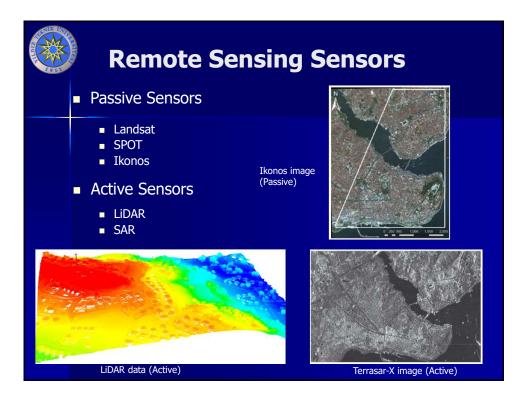
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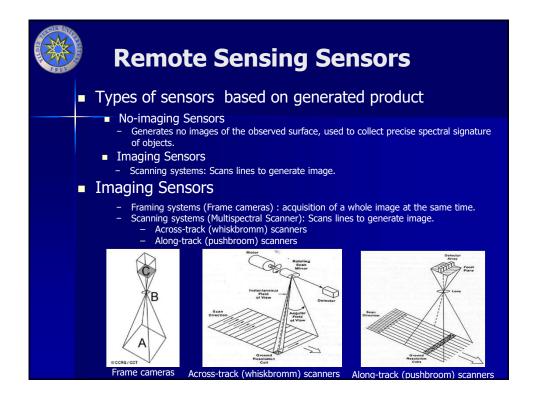
Spacing Between Adjacent Landsat 5 or 7 Orbit Tracks at the Equator

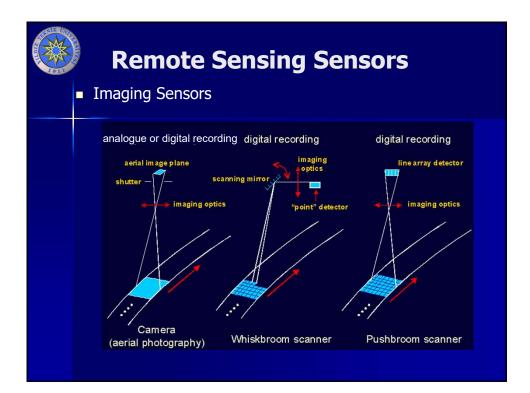










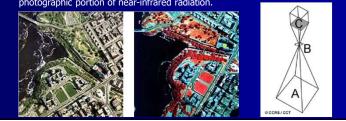


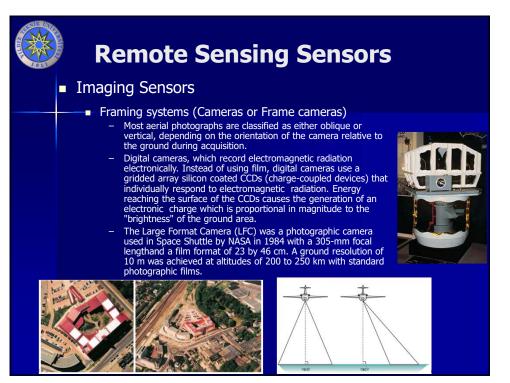
Remote Sensing Sensors

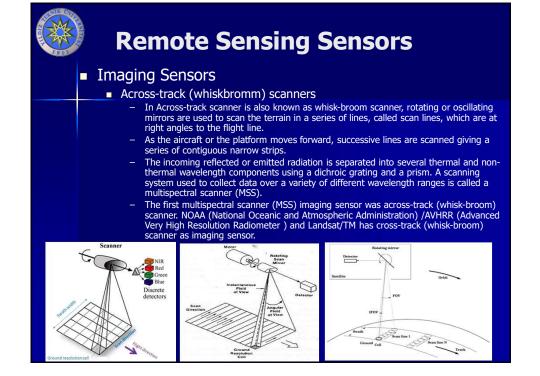
Imaging Sensors

- Framing systems (Cameras or Frame cameras)
 - Cameras (frame cameras) and their use for aerial photography are the simplest and oldest of sensors used for remote sensing of the Earth's surface.

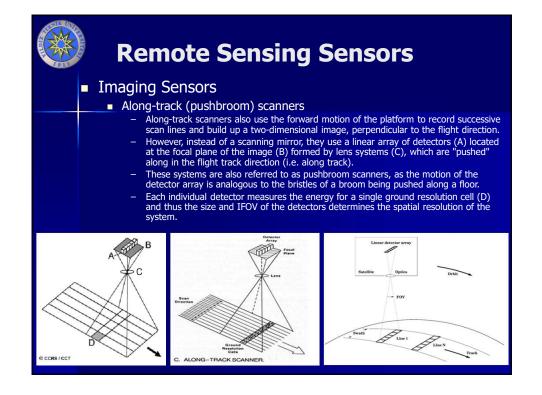
 - Cameras are framing systems which acquire a near-instantaneous "snapshot" of an area (A), of the surface. Camera systems are passive optical sensors that use a lens (B) to form an image at the focal plane (C). Photographic films are sensitive to light from 0.3 µm to 0.9 µm in wavelength covering the ultraviolet (UV), visible, and near-infrared (NIR). Panchromatic films are sensitive to the UV and the visible portions of the spectrum.
 - Colour and false colour (or colour infrared, CIR) photography involves the use of a three layer film with each layer sensitive to different ranges of light.
 - For a normal colour photograph, the layers are sensitive to blue, green, and red light. In colour infrared (CIR) photography, the layers are sensitive to green, red, and the photographic portion of near-infrared radiation.

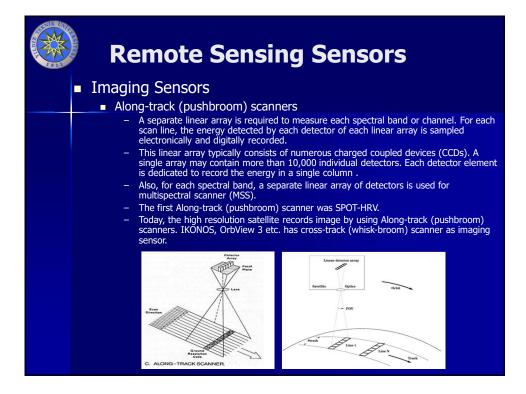


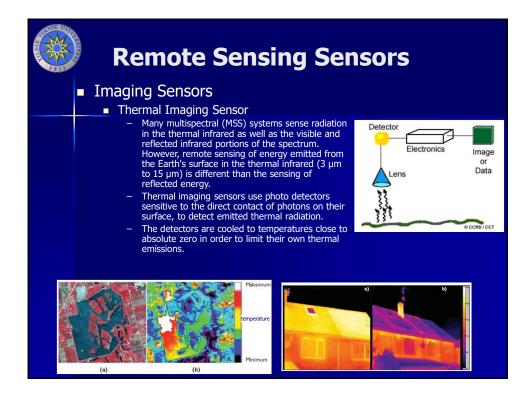


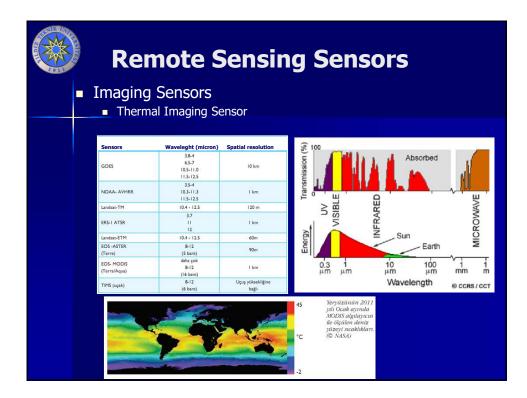


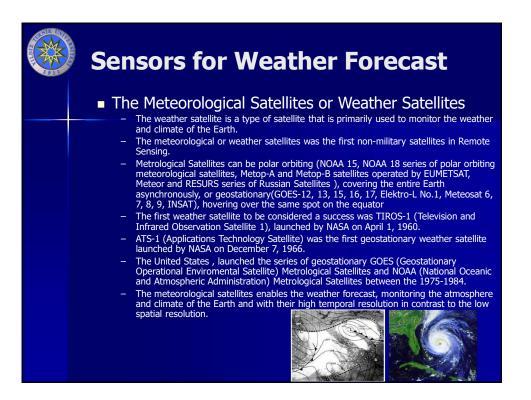
	Remote Sensing Sensors
∎ In	naging Sensors
	 Across-track (whiskbromm) scanners The IFOV is the angular cone of visibility of the sensor (A) and determines the area on the Earth's surface which is "seen" from a given altitude at one particular moment in time (B). The size of the area viewed is determined by multiplying the IFOV by the distance from the ground to the sensor (C). The detail discernible in an image is dependent on the spatial resolution of the sensor. The spatial resolution is the size of the smallest possible feature that can be detected in satellite image and depends primarily on their Instantaneous Field of View (IFOV) of passive sensors.
	$FOV=B=A^*C$

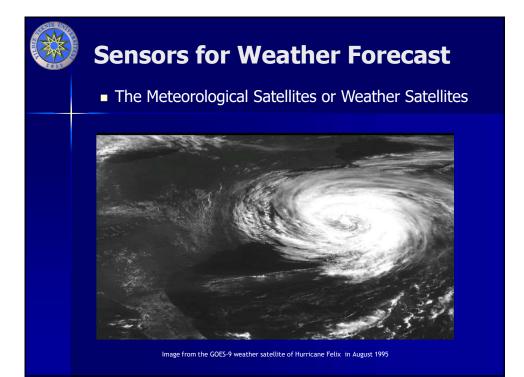












Se	ens	sors	s fo	r \	Ne	atl	ıeı	۰F	ore	ecast	
 GEOS (Geostationary Operational Environmental Satellite) 											
 The Geostationary Operational Environmental Satellite system (GOES), operated by th United States' National Oceanic and Atmospheric Administration (NOAA) supports weather forecasting, severe storm tracking, and meteorology research. The GOES system uses geosynchronous satellites and operate in geostationary orbit 35,790 (approximately 36,000) kilometers. 											
	Designatio	on Operational	Launch Date/Time (UTC)	Rocket	Launch Site	Longitude	First Image	Status	Retirement	Remarks	
		chronous Meter ured by <u>Ford Ae</u>	-	lites), -der	ived satellite	s[<u>edit]</u>					
	GOES-A	GOES-1	October 16, 1975, 22:40	<u>Delta</u> 2914	CCAFSLC- 17A		October 25, 1975	Retired	March 7, 1985 ^[12]		
	GOES-B	GOES-2	June 16, 1977, 10:51	<u>Delta</u> 2914	CCAFSLC- 17B	60°W		Retired	1993 ^[13]	Reactivated as comsat in 1995, ^[13] finally deactivated in May 2001	
	GOES-C	GOES-3	June 16, 1978, 10:49	<u>Delta</u> 2914	CCAFSLC- 17B			Retired	1993 ^[14]	Reactivated as comsat in 1995, ^[14] decommissioned 29 June 2016	
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Sensors f	or Wa	ather	Forecast
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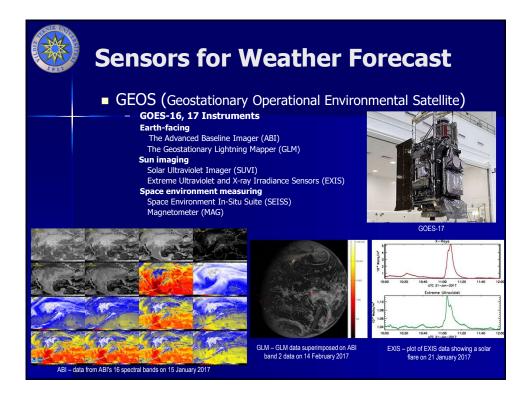
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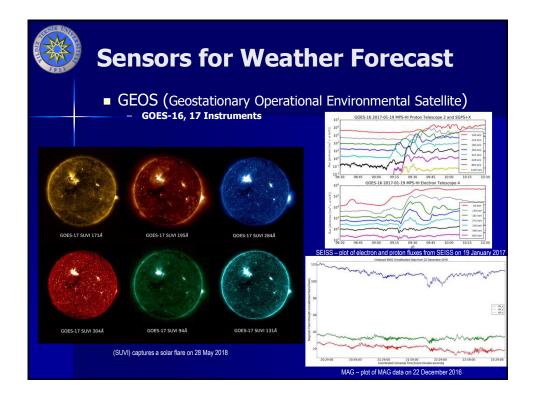
Designation		Dealers	Launch		First			Remarks			
Operational	Date/Time (UTC)	Rocket	Site	Longitude	Image	Status	Retirement	Remarks			
First generation[<u>edit]</u> Built on a <u>Hughes Space and Communications</u> <u>HS-371</u> spacecraft bus											
GOES-4	September 9, 1980, 22:57	<u>Delta</u> <u>3914</u>	CCAFSLC- 17A	135°W		Retired	November 22, 1988 ^[15]				
GOES-5	May 22, 1981, 22:29	Delta 3914	CCAFSLC- 17A	75°W		Retired	July 18, 1990 ^[16]				
GOES-6	April 28, 1983, 22:26	<u>Delta</u> <u>3914</u>	CCAFSLC- 17A	136°W ^[17]		Retired	January 21, 1989 ^[17]				
N/A	May 3, 1986, 22:18	<u>Delta</u> 3914	CCAFSLC- 17A	135°W (planned)	N/A	Failed	+71 seconds	Launch failure ^[18]			
<u>GOES-7</u>	February 26, 1987, 23:05	<u>Delta</u> <u>3914</u>	CCAFSLC- 17A	75°W, 98°W, 112°W, 135°W, 95°W, 175°W		Retired	January 1996 ^[19]	Reactivated as comsat for <u>Peacesat</u> from 1999- 2012, moved to <u>graveyard</u> <u>orbit</u> April 12, 2012. ^[20]			
	Operational ration[edit] Hughes Space a GOES-4 GOES-5 GOES-6 N/A	Detection Date/Time (JUTC) ration(edit) UTC) Hughes Space and Communicat September 9, 1980, 22:57 GOES-4 September 1981, 22:20 GOES-5 May 22, 1981, 22:20 GOES-6 April 28, 1983, 22:26 N/A May 3, 1986, 22:18 GOES-7 February 26,	Determine (UTC) Pocket (UTC) Operational stational Hughes Space and Communications HS-37 GOES-4 September 9, 1980, 22:57 Petra 3914 GOES-5 May 22, 1981, 22:29 Delta 3914 GOES-6 April 28, 1983, 22:26 Delta 3914 N/A May 3, 1986, 22:18 Delta 3914 GOES-7 February 26, 20:57 Delta	Determine Rocket Launch Ste Operational traionediti Variane Ste Hughes Space and Communications HS-371 spaceraft September 3914 Delta 3914 CCAFSLC 172 GOES-4 May 22, 1981, 22:29 Delta 3914 CCAFSLC 172 GOES-6 May 3, 1986, 1983, 22:26 Delta 3914 CCAFSLC 172 N/A May 3, 1986, 22:18 Delta 3914 CCAFSLC 172 GOES-7 February 26, Delta Delta 272 CCAFSLC	Operational (UTC) Date/Time (UTC) Rocket Launch Site Longitude ration[edit]: Hughes Spacea ad Communications HS-371 spacecraft September 9, 1980, 22:57 Delta 3914 CAFSLC: 17A 135°W GOES-4 September 9, 1980, 22:57 Delta 3914 CAFSLC: 17A 135°W GOES-5 May 22, 1981, 22:29 Delta 3914 CAFSLC: 17A 75°W GOES-6 April 8, 22:26 Belta 3914 CAFSLC: 17A 136°W ^{UTL} N/A May 3, 1986, 20:18 CAFSLC: 17A 136°W ^{UTL} GOES-7 February 26, 1987, 23:05 Delta 3914 CAFSLC: 17A 135°W, 135°W, 95°W, 135°W, 95°W, 135°W, 95°W,	Determinant Date/Time (UTC) Rocket Launch Ste Longitude First Image ration[edit] Hughes Space and Communications H5:371 spacecraft bss bss	Operational (urc) Date/ Rocket Jaunch Site Longitude First Image Status ration(edit: Hughes Space and Communications HS-371 space and HS-371	Operational (urc) Date / Time (urc) Rocket Launch Ste Longitude First mage Status Retirement ration(edit): Hughes Space and Communications HS-371 spacecraft GOES-4 September 9, 1980, 22:25 Defta CCAFSLC: 3214 135°W Retired Retired Jonember 2, 1988 GOES-6 May 2, 2 1983, 22:26 Defta CCAFSLC: 3214 136°W I.M Retired January 2, 1 1989, 22:45 GOES-6 May 3, 1986, 2914 CAFSLC: 3214 136°W N/A Faled +71 seconds GOES-7 February 26, 1924 Defta CAFSLC: 121°W, 127W, 137W, 95°W, 135°W, 95°W, 135°W, 135°W			

Sensors for Weather Forecast Geostationary Operational Environmental Satellite Server States' National Creational Environmental Satellite system (GOES), operated by the United States' National Oceanic and Atmospheric Administration (NOAA) supports weather forecasting, severe storm tracking, and meteorology research. The GOES system uses geosynchronous satellites and operate in geostationary orbit 37,790 (approximately 36,000) kilometers.											
	Designati Launch	on Operational	Launch Date/Time (UTC)	Rocket	Launch Site	Longitude	First Image	Status	Retirement	Remarks	
		eneration[edit] Space Systems/	' <u>Loral LS-1300</u> s	pacecraft	bus						
	GOES-I	GOES-8	April 13, 1994, 06:04	<u>Atlas I</u>	CCAFSLC- 36B	75°W	May 9, 1994	Retired	May 4, 2004 ^[21]	In graveyard orbit	
	GOES-J	GOES-9	May 23, 1995, 05:52	<u>Atlas I</u>	CCAFSLC- 36B	135°W, 155°E	June 19, 1995	Retired	June 14, 2007 ^[22]	In graveyard orbit	
	GOES-K	GOES-10	April 25, 1997, 05:49	<u>Atlas I</u>	CCAFSLC- 36B	135°W, 65°W	May 13, 1997	Retired	1 December 2009 ^[23]	In graveyard orbit	
	GOES-L	<u>GOES-11</u>	May 3, 2000, 07:07	Atlas IIA	CCAFSSLC- 36A	135°W	May 17, 2000	Retired	16 December 2011 ^[24]	Retired, Drifting west	
	GOES-M	<u>GOES-12</u>	July 23, 2001, 07:23	<u>Atlas</u> <u>II</u> A	<u>CCAFSSLC-</u> <u>36A</u>	60°W	August 17, 2001	Retired	August 16, 2013	Operated at GOES-South covering South America, and retained as spare, following replacement at GOES-East by GOES-13. Now in a <u>graveyard</u> <u>orbit</u> .	

 Sensors for Weather Forecast GEOS (Geostationary Operational Environmental Satellite) The Geostationary Operational Environmental Satellite system (GOES), operated by United States' National Oceanic and Atmospheric Administration (NOAA) supports weather forecasting, severe storm tracking, and meteorology research. The GOES system uses geosynchronous satellites and operate in geostationary orbit 35,790 (approximately 36,000) kilometers. 											
_	esignatio aunch	on Operational	Launch Date/Time (UTC)	Rocket	Launch Site	Longitude	First Image	Status	Retirement	Remarks	
		eration[edit] Boeing BSS-601	spacecraft bus					1			
G	OES-N	<u>GOES-13</u>	May 24, 2006, 22:11	<u>Delta IV-</u> <u>M+(4,2)</u>	CCAFSSLC- <u>37B</u>	75°W	June 22, 2006	Standby		Replaced by <u>GOES-16</u> at GOES- East on December 18, 2017. Instruments were shut off on January 2, 2018 and began a drift to a storage position at 60 degrees west longitude, where it may take over GOES-South dutles. ^[22]	
G	OES-O	GOES-14	June 27, 2009, 22:51	<u>Delta IV-</u> <u>M+(4,2)</u>	<u>CCAFSSLC-</u> <u>37B</u>	105°W	27 July 2009	Standby		On-orbit spare, was used to cover GOES-East imagery and moved into position following GOES-13 malfunction in 2012. ^[22] also activated to cover GOES-13 outage in mid-2013	
G	OES-P	<u>GOES-15</u>	4 March 2010, 23:57	<u>Delta IV-</u> <u>M+(4,2)</u>	CCAFSSLC- 37B	89.5°W, 135°W	7 April 2010	Active ^[27]		Still active, but since replaced operationally by GOES-17 as GOES-West	

Sen	sor	s for	- M	lea	th	er	Fo	ore	cast	
 GEOS (Geostationary Operational Environmental Satellite) GOES-17 (formerly GOES-S) satellite was launched into space on 1 March 2018 in geosynchronous orbit at 35,700 km above Earth by an Atlas V (541) vehicle from Ca Canaveral Air Force Station, Florida. GOES-17 has the same instruments and capabilities as GOES-16 (formerly GOES-East These two satellites are expected to monitor most of the Western Hemisphere and detect natural phenomena and hazards in almost real time GOES-16, 17 Instruments Earth-facing										
Designati Launch	on Operational	Launch Date/Time (UTC)	Rocket	Launch Site	Longitude	First Image	Status	Retirement	Remarks	
Schedule	d launches									
	neration (GOES Lockheed Marti	-R Series)[edit] n A2100 spacecraf	't bus							
GOES-R	GOES-16	19 November 2016, 23:42 ^[29]	Atlas V 541	CCAFSSLC- 41	89.5°W	January 15, 2017	Active		Replaced <u>GOES-13</u> at GOES- East on December 18, 2017. ^{[25][30]}	
GOES-S	GOES-17	March 1, 2018 ^[31]	Atlas V 541	CCAFSSLC- 41	137°W		Active		GOES-West	
GOES-T		2020 ^[32]	EELV	CCAFS						
GOES-U		2024	<u>EELV</u>	CCAFS						





 Sensors for Weather Forecomental GEOS (Geostationary Operational Environmental GOES-16, 17 Instruments 	Satellite)
Pixel depending on t	the type of image the western hemisphere
$\underline{\lambda}(\mu m)$ $\underline{\lambda}(\mu m)$ (km)	5 to 15 minutes, while
	s was a scheduled event, three photos per hour.
	e continental United States
	minutes, compared to one
	utes in previous satellites
5 1.58–1.64 1.61 1 Snow/Ice Near-Infrared Snow/Ice discrimination, 1000 km box	mage over some 1000 by every thirty seconds, a vious imagers did not have
6 2.225-2.275 2.25 2 Cloud Particle Size Near-infrared Cloud particle size, snow ABI spatial res	
7 3.80-4.00 3.90 2 Shortwave Window Infrared Fog. stratus, fire, band 2 is the	highest resolution out of all a resolution of 500 m
	, and 5 will have a resolution
9 6.75–7.15 6.95 2 Mid-level Tropospheric Water Vapor Vapor GLM is used for r and cloud-to-gro	measuring lightning (in-cloud ound) activity
10 7.24–7.44 7.34 2 Lower-level Tropospheric Infrared Water vapor features – GLM records a	a single channel in the NIR
	flashes from lightning.
12 9.42–9.8 9.61 2 Ozone Infrared Total column ozone – GLM has a 13	72×1300 pixel CCD, with an
13 10.1–10.6 10.35 2 Window Infrared Clouds resolution dec	al resolution (with the creasing near the edges of
14 10.8–11.6 11.2 2 Infrared Longwave Window Infrared Clouds The FOV).	a frame rate of 2
15 11.8–12.8 12.3 2 Dirty Infrared Longwave Infrared Clouds milliseconds, r	meaning it considers the rea 500 times every second.
16 13.0–13.6 13.3 2 CO2 Longwave Infrared Infrared Air temperature, clouds	



Sensors for Weather Forecast

- NOAA (National Oceanic and Atmospheric Administration) AVHRR (Advanced Very High Resolution Radiometer)
 - NOAA has at least two polar-orbiting meteorological satellites in orbit at all times, with one satellite crossing the equator in the early morning and early evening and the other crossing the equator in the afternoon and late evening.
 - The primary sensor on board both satellites is the AVHRR instrument. The AVHRR is a radiation-detection imager that can be used for remotely determining cloud cover and the surface temperature. The term surface can mean the surface of the Earth, the upper surfaces of clouds, or the surface of a body of water.
 - Morning-satellite data are most commonly used for land studies, while data from both satellites are used for atmosphere and ocean studies. Together they provide twice-daily global coverage, and ensure that data for any region of the earth are no more than six hours old.

 - The swath width, the width of the area on the Earth's surface that the satellite can "see", is 2,343 km from the 833 km nominal orbital altitude. The satellites orbit between 833 or 870 kilometers (+/- 19 kilometers above the surface of the Earth. The first AVHRR was a 4-channel radiometer, first carried on TIROS-N (launched October 1978). This was subsequently improved to a 5-channel instrument (AVHRR/2) that was initially carried on NOAA-7 (launched June 1981). The latest instrument version is AVHRR/3, with 6 channels, first carried on NOAA-15 launched in May 1998.

Sensors for Weather Forecast

- NOAA (National Oceanic and Atmospheric Administration) AVHRR (Advanced Very High Resolution Radiometer)
 - Measuring the same view, this array of diverse wavelengths, after processing, permits multi spectral analysis for more precisely defining hydrologic, oceanographic, and meteorological parameters.

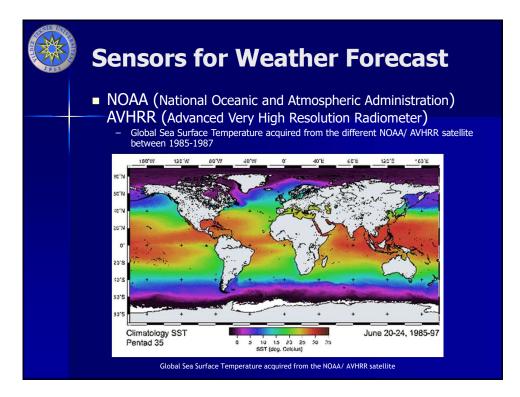
AVHRR/3 Channel Characteristics								
Channel Number	Resolution at Nadir	Wavelength (um)	Typical Use					
1	1.09 km	0.58 - 0.68	Daytime cloud and surface mapping					
2	1.09 km	0.725 - 1.00	Land-water boundaries					
ЗА	1.09 km	1.58 - 1.64	Snow and ice detection					
3B	1.09 km	3.55 - 3.93	Night cloud mapping, sea surface temperature					
4	1.09 km	10.30 - 11.30	Night cloud mapping, sea surface temperature					
5	1.09 km	11.50 - 12.50	Sea surface temperature					

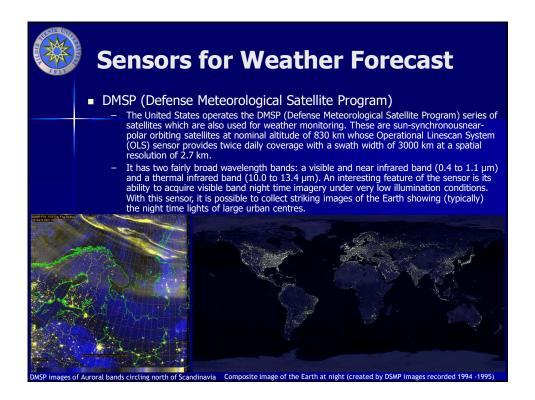


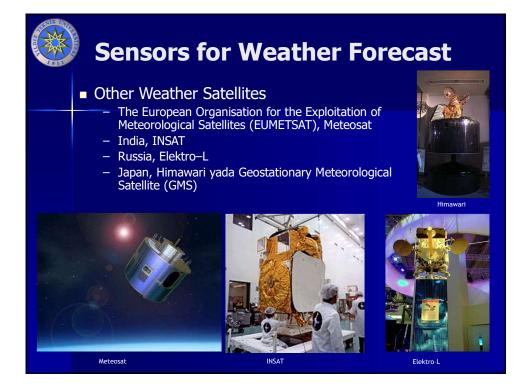
Sensors for Weather Forecast

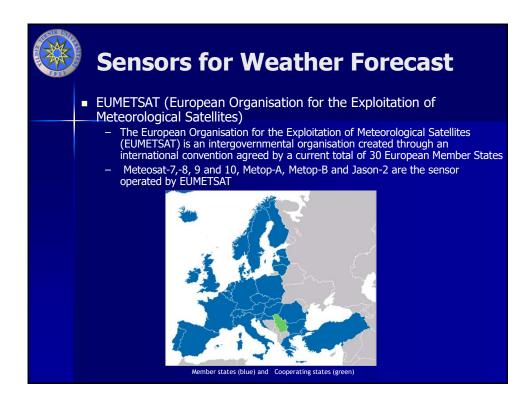
NOAA (National Oceanic and Atmospheric Administration AV/HPR (Advanced Vor Ulink Resolution Rediameter))							
 AVHRR (Advanced Very High Resolution Radiometer) NOAA Satellite Launch and Service Dates 								

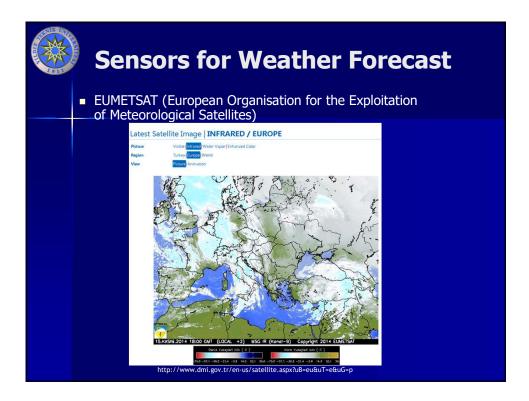
Satellite name	Launch date	Service start	Service end		
TIROS-N ['tairəus] [Television and Infrared Observation Satellite]	13 October 1978	19 October 1978			
NOAA-6	27 June 1979	27 June 1979	16 November 1986		
NOAA-7	23 June 1981	24 August 1981	7 June 1986		
NOAA-8	28 March 1983	3 May 1983	31 October 1985		
NOAA-9	12 December 1984	25 February 1985	11 May 1994		
NOAA-10	17 September 1986	17 November 1986	17 September 1991		
NOAA-11	24 September 1988	8 November 1988	13 September 1994		
NOAA-12	13 May 1991	14 May 1991	15 December 1994		
NOAA-14	30 December 1994	30 December 1994	23 May 2007		
NOAA-15	13 May 1998	13 May 1998	Present		
NOAA-16	21 September 2000	21 September 2000	9 June 2014		
NOAA-17	24 June 2002	24 June 2002	10 April 2013		
NOAA-18	20 May 2005	30 August 2005	present		
NOAA-19	6 February 2009	2 June 2009	present		

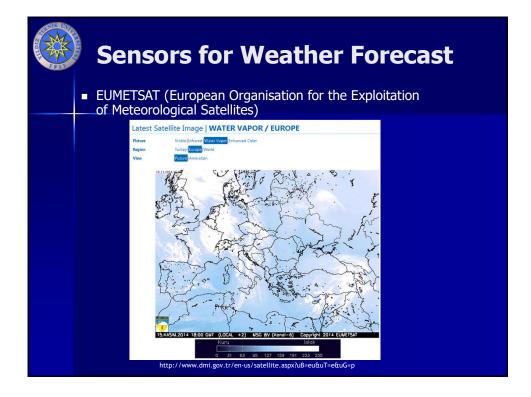


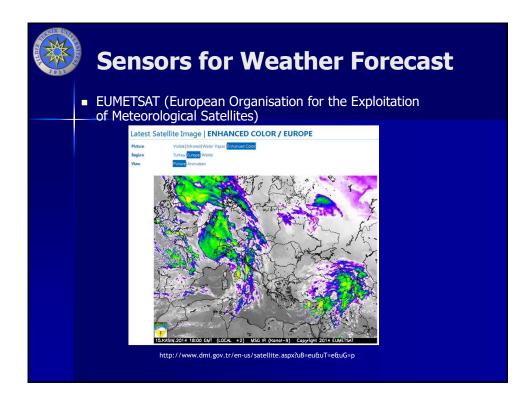


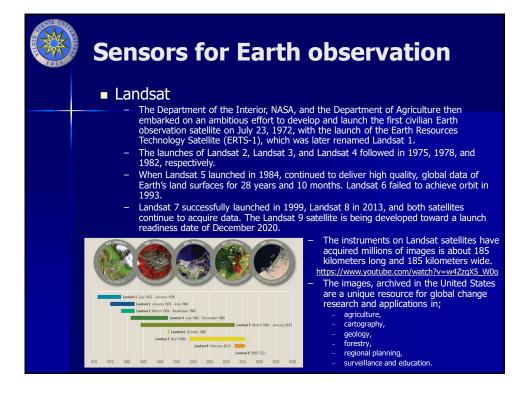












Sen	SO	rs f	for	·E	arth observati	on
Lar						
	esat 1	• Launched •	January 6, 1978	5 years, 6 months and 14 days	Know Originally avaid Earth Resources Technology Skellites 1. Landaut 1. carried to vir kill internements: a currents built by the Radio Composition of American (EAC) known as the Returns Bearn Vidson (PKP). Aid well as a Multi spectral Scenner (MSS) built by the Higher Aircraft Company.	
Lan	stat 2	January 22, 1975	February 25, 1982	7 years, 1 month and 3 days	Newly identical carpy of Landset 1. Payload consisting of a Baton Baron Vedeon (RBV) and a Multi- spectral Science (MSS). The specifications of these instruments were identical to Landset 1.	
Lan	trat 3	March 5, 1978	March 31, 1983	5 years and 26 days	Naviry identical copy of Landsat 1 and Landsat 2. Payload consisting of a Return Beam Videon (RBM) as well as a Multi spectral Scenare (MSS), included with the MSS was a short-lived thermal band MSS data was considered more scenificity applicable than the RBV which was neety used for engineering evaluation purposes.	
Lan	stat 4	July 16, 1982	December 14, 1993	11 years, 4 months and 28 days	Landsat 4 carried an updated Multi Spectral Scanner (MSS) used on previous Landsat missions, as well as a Thematic Mapper.	
Lan	5 at 5	March 1, 1984	June 5, 2013 ⁽¹⁰⁾	29 years, 3 months and 4 days	Nearly identical copy of Landsat 4. Longest Earth-observing satellite mission in history. Designed and built at the same time as Landsat 4, this satellite camed the same payload consisting of a Multi Spectral Scanner (MSS) as well as a Thematic Mapper.	
Lan	faat 6	October 5, 1993	October 5, 1993	0 days	Failed to reach obit. Landsat 6 was an upgraded version of its predecessors. Carrying the same Multi spectral Scanner (MSS) but also carrying an Enhanced Thematic Mapper, which added a 15m- resolution panchromatic band.	
Lan	faat 7	April 15, 1999	Still active	19 years, 11 months and 12 days	Operating with scan line connector disabled since May 2003 ¹¹⁰ The main component on Landsat 7 was the Exhanced Thematic Mapper Plus (ETM-) Stati consisting of the 15m-esolution parcheomatic band, but also includes a Kill spintrue calibration. This allows for 5% absolute radiometric calibration. ¹¹⁰	
Lanv	esat 8	February 11, 2013	Still active	6 years, 1 month and 16 days	Originally named Landard Data Costinuity Mission from layerch and May 30, 2013, when NASA generations were transfer over to USSIC 97 ¹⁰ Landard that two sensions with its psychold, the Operational Land Images (OL) and the Thermal United Sensor (RHS) ¹¹⁴	
Lan	feat 9	December 2020 (expected)			Landsat 9 will be a rebuild of its predecessor Landsat 8 (¹⁶⁾	

	I dsat Landsat 1 through 5 Landsat 4 and 5 car						ruments.
		Landsat 1-5 Multis	pectral S	Scanner (N	ISS)		
	Landsat 1-3 MSS	Landsat 4-5 MSS Wa		Wavelength (micrometers)		Resolution (meters)	
	Band 4 - Green	Band 1 - Green	0.5 - 0.6 0.6 - 0.7			60*	
	Band 5 - Red	Band 2 - Red			0	60*	
	Band 6 - Near Infrared (NIR)	Band 3 - NIR 0.7 -				60*	
	Band 7 - NIR	Band 4 - NIR	60*				
Landsat -	4-5 Thematic Mapper (TM) Wavelength (micrometers	s) Resolution (me	eters)	Channel TM 1	Wavelength Range (µm)	coll/uppotation disoriminat	lication
Band 1 - Blue	0.45 - 0.52	30		0.52 0.60		(e) mapping; cultural/urban feature identification green vegetation mapping (measures reflectance	
Band 2 - Green	0.52 - 0.60	30		TM 2	(green)	peak); cultural/urban feati	ure identification
Band 3 - Red	0.63 - 0.69	30		TM 3	0.63 - 0.69 (red)	vegetated vs. non-vegetated and plant species discrimination (plant chlorophyll absorption); cultural/urban feature identification	
Band 4 - NIR	0.76 - 0.90			TM 4	0.76 - 0.90 (near	r identification of plant/vegetation types, health biomass content; water body delineation; soi	
Band 5 - Shortwave Infrared (SWIR) 1	1.55 - 1.75	30		TM 5	1.55 - 1.75 (short wave IR)	t sensitive to moisture in soil and vegetation; discriminating snow and cloud-covered areas	
Band 6 - Thermal	Band 6 - Thermal 10.40 - 12.50			TM 6	10.4 - 12.5	vegetation stress and soil	moisture discrimination
Band 7 - SWIR 2	2.08 - 2.35	30		(M O	(thermal IR)	water)	n; thermal mapping (urban,
				TM 7	2.08 - 2.35 (short wave IR)	discrimination of mineral a vegetation moisture conte	and rock types; sensitive to

Sensors for Earth observation

Landsat

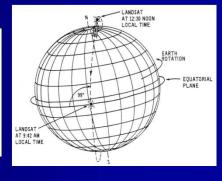
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- Landsat 7 uses the Enhanced Thematic Mapper Plus (ETM+) scanner.

- Landsat 7 uses the Linnanced memauc Mapper Plus (ETM+) scanner. Landsat 8 uses two instruments, the Operational Land Imager (OLI) for optical bands and the Thermal Infrared Sensor (TIRS) for thermal bands. Landsat missions use sun-synchronous, near polar orbits at different altitudes for each mission.

https://www.youtube.com/watch?v=xBhorGs8uy8

Landsat 7 Enhanced Thematic Mapper Plus (ETM+)								
Bands	Wavelength (micrometers)	Resolution (meters)						
Band 1 - Blue	0.45 - 0.52	30						
Band 2 - Green	0.52 - 0.60	30						
Band 3 - Red	0.63 - 0.69	30						
Band 4 - NIR	0.77 - 0.90	30						
Band 5 - SWIR 1	1.55 - 1.75	30						
Band 6 - Thermal	10.40 - 12.50	60* (30)						
Band 7 - SWIR 2	2.09 - 2.35	30						
Band 8 - Panchromatic	0.52 - 0.90	15						



Sensors f	or Ea	rth ob	servat	tion

		Lar	nds	at																
Mission	Lands	at-1	Lands	at-2	Lands	at-3	Landsat-	4	Landsat-	5	Landsat-6	Landsat-7	Landsat-	8(LCDM)						
Mission period	1	972-1978	1	975-1982	1978-1983		19	82-2001	1984-2012		1993, failed	April 1999 -	Feb 2013							
Orbit							Sur	Sun-synchronous, near-polar												
Altitude		917 km		917 km		917km		705 km		706 km		705kn	1	705 km						
Inclination		99.2 deg		99.2 deg		99.2 deg	. 8	98.2 deg	9	98.2 deg		98.2deg	5	98.2 de						
Eq. crossing (+/- 15min)		9:30am		9:30am 9:30am		9:45am		9:45am		1 0an	n.	10 ar								
Period (min)		103.34		103		103		99				98.9	9 9							
No. orbits /day		14		14		14		14		14		14		14		14		14		14
Repeat cycle		18		18		18		16		16		16	5	1						
Swath width		185 185 185		185			185		185		18									
Sensors	RBV	MSS	RBV	MSS	RBV	MSS	MSS	TM	MSS	TM	ETM	ETM+	OLI	TIRS						
Bands	1-3	4-7	1-3	<mark>4</mark> -7	1-4	4-8	1-4	1-7	1-4	1-7	1-8	1-8	1-9	1-2						
Spatial resolution (m)	80	82	80	82	80	82 B8:240	79	30 B6:120	79	30 B6:120		B1-B5,B7: 30 B6: 60 B8: 15	30 B8:15	100						
Radiometric resolution (Bits)	6	B1-B3:7 B4: 6	6	B1-B3:7 B4: 6		B1-B3:7 B4: 6	B1-B3:7 B4: 6		B1-B3:7 B4: 6	8	8	8	12	12						

